

TESTING OF MECHANICAL PROPERTIES OF ALUMINIUM-LITHIUM (Al-Li) 8090 WITH REINFORCEMENT OF SILICON CARBIDE (SiC)

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Abstract *The utilization of Aluminium composites in the defense and aerospace industries has skyrocketed recently. Due to their low density combined with high strength and stiffness, aluminum-lithium (Al-Li) based composites have increased in popularity in recent years. Lithium (Li) can be used to reduce density while also increasing strength and stiffness. Despite this, aluminum-lithium (Al-Li) composites have gotten comparatively little research. Silicon Carbide (SiC) particles were found to be an outstanding reinforcement with the Aluminium-Lithium (Al-Li) matrix among the numerous ceramic reinforcements tested. The addition of Silicon Carbide (SiC) particles to the Aluminum matrix increases the material's strength and young's modulus, hence increasing its particular qualities. In comparison to the unreinforced alloy, the Aluminium-Lithium-Silicon Carbide (Al-Li-SiC) composite has superior mechanical properties. Stir Casting Technique can be used to determine Mechanical Properties such as Tensile Strength, Compressive Strength, Hardness, and Wear Resistance of Aluminium- Lithium-Silicon Carbide (Al-Li-SiC) composite.*

1. INTRODUCTION

For the rising new technology, we have a huge requirement for materials to process everything with particular features. Traditional materials, on the other hand, are unable to match these unique features, such as a high strength-to-weight ratio. To achieve these unique qualities, we create new materials by combining two or more insoluble elements, known as composites. In today's world, hybrid metal matrix materials are crucial in fields such as mechanical, electrical, car, aerospace, material science, and marine engineering.

The primary goal of the HMMC preparation was to improve mechanical and thermal qualities such as tensile strength, hardness, wear resistance, thermal conductivity, and melting point. Hybrid Metal Matrix Composites is gearing up to meet engineering demands at the lowest possible cost. Advanced composites have outperformed many other materials in a variety of applications, including tennis racquets, industrial rollers, and space antennae. HMMCs are used to process almost all consumer and engineering applications. In comparison to other metal matrix composites, aluminium metal matrix composites (AMCs), lithium metal matrix composites (Li-Alloys), and lithium metal matrix composites (Li-Alloys) have a low density value. In the sphere of engineering, these AMCs are primarily used as a supplement to high-density materials like steels. Fuel and maintenance costs, as well as material cost. Sand casting has been used to manufacture metal components from ancient times (4000 B.C.), and turning is the most basic metal cutting process for preparing end components with a small number of control parameters and a cheap cost of production.

The matrix components in Metal Matrix Composites are metals, and the reinforcing materials are ceramics, polymers, or natural materials. Cast Aluminium-alloys, Iron, Steels, Ti-alloys, Cu-alloys, and Ni-alloys are some examples of MMCs. These have better mechanical, thermal, and electrical capabilities than the base materials, allowing them to meet the needs of the particular location after composition. When we mix up Ceramics like Silicon Carbides or Alumina or Silica or Boron Carbides or Silicon Carbide up to 40% weight of reinforcing material mixed up to base material through the various processing processes of HMMCs at optimum cost basis, the Aluminium Metal Matrix Composites have a lower density than the base material. The density values of the base material are easily lost during this procedure. The reason for this is that ceramics have a lower density than metals.

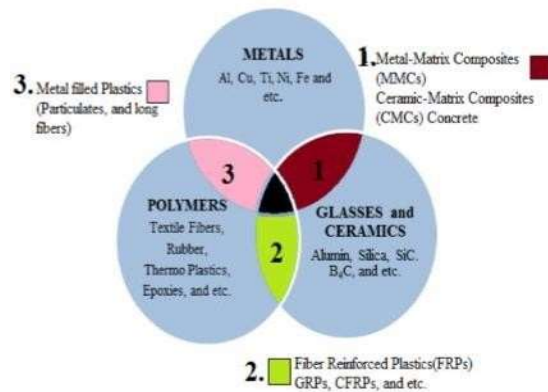


Fig1:DescriptionofCompositeMaterials

The process of choosing a material for a certain purpose is an interesting one. However, this is largely determined by factors such as need requirements, fabrication processes, cost, and final object functioning conditions. The continual desire for greater fuel efficiency in the region's and vehicle sectors has resulted in the emergence of low-density Al-Li alloys. 1 Chronicles additions Li will reduce the density by a factor of three, with a corresponding gain in strength and stiffness. To gain an additional boost in strength and stiffness, it makes sense to consider reinforcing Al-Li alloys with ceramic particles. Despite some recent research on Al-Li alloy-based composites, Al-Li alloy-based composites have gotten relatively little attention. This could be due to the dangers of handling Li during alloying. To prevent Li loss and the related fire hazard/burning at the process temperature, a meticulously controlled environment or vacuum is required, all of which results in expensive processing/material prices. By altering the usual stir casting procedure for the manufacture of Al-Li-SiC composites, a simple and efficient experimental setup was adopted in this study. Additionally, the mechanical properties of assail particle-dispersed Al-Li-SiC composites, as well as their age-hardening mechanics, have been investigated.

1.1 ALUMINIUMALLOY

Aluminium is the most plentiful metal in the earth's crust, accounting for around 8.1 percent of all metals in the form of oxides. We were unable to obtain any metal in the form of our desired base pure metal from the earth crust or natural sources. Alumina is an oxide that occurs naturally in ruby, sapphire, corundum, and emery. Today, however, the majority of our aluminium comes from a synthetic blend of sodium, aluminium, and calcium fluorides. Clay can also be used to make aluminium, although the technique is currently uneconomical.

Table1:PropertiesofPureAluminium

Densit	2.7 g/cm ³
Melting Point	658 ⁰ c
Thermal Capacity	900 J/Kg
Thermal Conductivity	230W/m
Coefficient of linear expansion	24×10 ⁻⁶ / ⁰ c
Electric Conductivity	60% as per I.A.C.S.
Electric Resistance	29×10 ⁻⁹ Ωm
Modulus of elasticity	70Pa

The unique qualities of atomic number 13 and its alloys make it one of the most flexible, cost-effective, and interesting antimonies materials for a wide range of applications, from soft, exceptionally ductile wrapping foil to the most stern engineering applications. In terms of structural metals, atomic number 13 alloys are second only to steels. It has a density of .7 g/cm³, which is nearly a third of the maximum amount of steel (7.83 g/cm³). A cu ft of steel weighs about 490 pounds, while a cu ft of atomic number 13 weighs just about 177 pounds. Because of their light weight, as well as the great strength of some atomic number 13 alloys (which exceeds that of structural steel), they can be styled and built in a variety of ways. light-weight constructions that are particularly beneficial for everything that moves—spacecraft and craft, as well as many types of land- and water-borne vehicles.

Mechanical Properties: The temper of the fabric has a big impact on the mechanical qualities of 8090

8090-8771

HeatMost enduringness is no more than 540MPa, and most yield strength is no more than 470MPa in heat treated 8090 (8090-8771temper). V-E Day elongation (stretch before final failure) is found in the fabric. It's incredibly corrosion-resistant and exceedingly strong.

8090-T3

T3 temper 8090 has a yield strength of 210MPa and a final enduringness of 340MPa (49,300psi) (30,500 psi). It's a thirteen failure elongation. T3 tempering is commonly accomplished by homogenizing the forged 8090 at 750 °C for several hours, quenching, and then ageing at 122 °C for twenty-four hours. This gives the 8090 alloy its maximum strength. The main source of strength is finely distributed eta and eta precipitates among grains and on grain borders.

8090-T8151

T651 temper 8090 has a Brinell hardness of 121 and an ultimate enduringness of 450MPa (65,300 psi). It's a seven-fold failure elongation. Depending on the shape of the fabric chosen, these qualities will change. Thicker plate may have lower strengths and elongation than the figures given above.

8090-T7

T81 temper has a yield strength of 340MPa and a final enduringness of 440MPa (63,800psi) (43,300psi). It's a thirteen failure elongation. Over aging (meaning ageing past the height hardness) the fabric produces the T81 temper. This is usually performed by ageing at 100–120 °C

1.2 SILICON CARBIDE

Silicon inorganic compound, often known as abrasive, is a unique compound of carbon and chemical element that is one of the toughest materials available. The molar mass of this substance is forty. The molecular weight is 10 g/mol. It's a simple compound in which each atom has a positive and electric charge and is connected to a chemical element via a triple bond. However, instead of being ionic, the bonding between them is valence. Solid carbide is found in a variety of crystalline shapes, with the polygonal shape crystal structure being the most commonly encountered occurrence. Moissanite, a rare mineral, produces silicon carbide naturally. Production carbide (SiC) is a man-made mineral that is most commonly produced in impedance furnaces using an industrial technique that was called for the Yankee E.G. Dean Acheson UN agency in 1891. A mixture of carbon material (typically oil coke) and an oxide or quartz sand is reacted with chemicals at high temperatures between 1700 and 2500°C in a Dean Acheson chamber, resulting in the creation of -SiC following the most reaction:

The reaction is powered by the resistive heating of a black lead core, which is achieved by connecting the core to two electrodes at opposite ends of the chamber. SiC forms as a solid cylindrical metal bar with radial layers that progress from black lead within the core to -SiC (the highest grade material with coarse crystalline structure), -SiC, scientific discipline grade, and finally unreacted material on the surface. Depending on the quality of the raw materials, these are frequently manufactured as either black or inexperienced materials. After a period of cooling, the assault metal bar is precisely separated and further treated for diverse applications. The resultant crude material is thoroughly crushed, categorised, usually polished again, and potentially chemically treated to obtain the specific qualities for which it will be used. The majority of our power and the worth we tend to increase our goods is accounted for by these ensuant procedure steps.



Fig2: Silicon Carbide Powder

2. EXPERIMENTAL PROCEDURE

Material selection and composite preparation:

Al-Li alloy (8090Al) is the metal alloy employed as the matrix material in this investigation. The experimental equipment for generating the composites might be a simple modification of the well-delineated conventional softens stirring approach. During the early stages. The stir casting setup used in this investigation is depicted in a schematic picture. The change is in the form of a steel cowl with an inert environment to prevent the loss of metallic elements. Previously, the material was stirred either in the open or in a chamber with a provision in earlier research.

During melting and stirring, this type of environment chamber prevents direct reading of the soften. This study's simple improvisation does not include It eliminates the need for maltreatment of such furnaces while also preventing Li loss. Frequently, the steel hood (gas cover) is elevated.



Fig3: Stir Casting Machine

2.1 METHOD OF FABRICATION

The matrix material was received in the form of a (950 gramme) ingot of aluminium alloy 8090. The ingots were chopped into smaller pieces, each weighing around 3 to 5 grains. This was done to reduce the alloy's dwell time, or to put it another way, to make it melt faster. Another advantage of having smaller pieces is that it makes it easier to get a more precise reading when weighing them. The ingots were chopped into small pieces using a cut off machine with a cutting wheel of grade HH, designed specifically for non-ferrous metals, and a continuous flow of coolant to prevent the ingot and cutting wheel from overheating. After cutting, the ingots were washed in warm water. Al 8090 in combination with using-Below is an example of an analytical balance using different weight percent's.

A J-type thermocouple was installed into the steel chamber to provide feedback of the temperature within the chamber to the furnace's temperature controller.

The nitrogen gas was set to flow continuously at a rate of 3cc/min, and the furnace was preheated to 100°C. All of the ingredients, including the aluminium alloy 8090 with SiC, were deposited in a graphite crucible. The crucible was then placed within the furnace in a stainless steel chamber, which was then sealed. The temperature of the furnace was gradually increased to reach 850°C in order to achieve a controlled temperature inside the chamber and inside the crucible (took about 4 hours period) With a ramp period of 2 hours, 1 hour, and 1 hour, this temperature was raised in increments to 200°C, 500°C, and 850°C.

2.2 COMPOSITE SPECIMEN PREPARATION

The experimental setup consists of the primary furnace and its components, as well as three mild steel stirrer blades. The experiment begins with a preheating procedure. Separately, the empty crucible and the SiC reinforcement powder are heated to a temperature close to the main

The graphite crucible is used to melt the aluminium alloy 8090(950gms) ingot within the furnace. The ingot was initially preheated for 4–5 hours at 560°C. Silicon carbide powder is also warmed to 300 degrees in the muffle furnace at the same time. The crucible with aluminium alloy is then heated to 840 degrees while the warmed powder is physically mixed with each other. In the furnace, the metal– matrix is then kept at the same temperature. In the furnace, the aluminium alloy particles and Silicon Carbide powder are completely melted. The stir-ring device is lowered into the crucible and adjusted to the correct depth within the furnace. The material is forcefully agitated at 700 rpm for 10 minutes, evenly dispersing the additive powder throughout the aluminium alloy matrix. The furnace temperature should be held at 830°F during the final mixing phase.

2.3 MECHANICAL TESTING SOF COMPOSITES:

A) TENSILE TEST

Metals, The tensile test is used to determine the strength of metals, wood, polymers, and most other materials. Tensile loads are those that rip the specimen apart and place it in tension. They can be performed on any specimen that can be subjected to a consistent tensile tension and has a known cross-sectional area and gauge length. Tensile tests are used to determine how a material will react when subjected to static, axial tensile, or stretch loads. ASTM standards for typical tensile tests can be found in Sections E8-2016a (metals), D638 (plastics), D2343 (fibres), D897 (adhesives), D987 (paper), and D412 (paper) (rubber).

PROCESS:

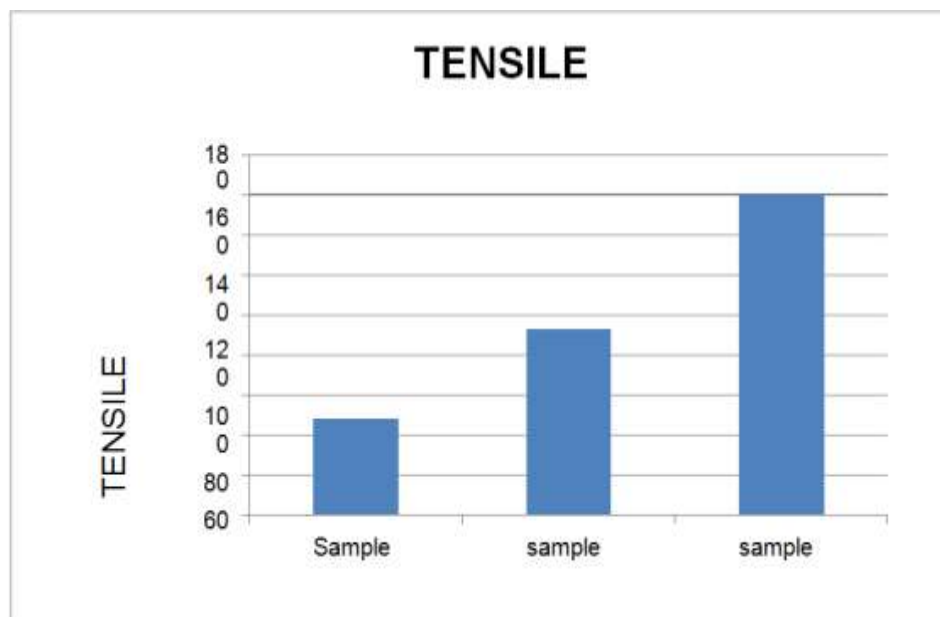
The tensile test is carried out using the "ASTM E8-2016a" technique.

- Tensile tests are used to determine the tensile properties of a material, such as its tensile strength. Tensile strength refers to a material's ability to withstand the most tensile stress possible. A suitable specimen must be collected before a tensile test can be performed. This specimen's size and characteristics should meet ASTM standards. Prior to the test, the cross-sectional area and a pre-determined gauge length can be calculated and indicated. The specimen is then placed in the proper grippers and introduced into a tensile load machine. After the machine has been loaded, it can be utilised to apply a steady, continuous pressure. Data is collected at specific intervals or locations throughout the test. Depending on the drug and specimen under examination, data points may be more or less frequent. The information includes both the applied load and the change in gauge length. The machine screen normally displays the load in pounds or kg.

To measure the change in gauge length, an extensometer is utilised. An extensometer, which is firmly attached to the machine or specimen, is used to determine the amount of deformation or deflection over the gauge length during a test. Data points are gathered while paying close attention to the readings until the material begins to show signs of yielding. This is obvious when deformation persists without the need to increase the applied load. At this time, the extensometer is removed, and loading is continued until failure. The ultimate tensile strength and rupture strength can be calculated using the latter loading.

Table2:Tensiletestvalues

SAMPLE	COMPOSITION OF COMPOSITE SPECIMEN	ULTIMATE TENSILE STRENGTH
SAMPLE-1	Al-950 gms + SiC-38 gms	48 Mpa
SAMPLE-2	Al-950 gms + SiC-76 gms	93 Mpa
SAMPLE-3	Al-950 gms + SiC-114 gms	160 Mpa

**Fig4:SamplesafterTensileTest****Fig5:GraphicalrepresentationofTensileTestvaluesonCompositesamples.**

B)HARDNESSTEST

Brinell hardness is determined by pressing a hard steel or carbide sphere of a specific diameter into a material's surface under a defined load and measuring the diameter of the indentation left behind. The end result is a pressure measurement, however the units are seldom given.

Penetration. As a result, the bulk of hardness tests include measuring the amount of force required to implant a specific indentation in a specimen's surface OR the size of the indentation produced by applying a given load.

Other hardness tests, such as sclera scope, necessitate the rebound of a dynamic or impact force, and the indenter used varies depending on the test. The amount of rebound achieved is utilised to calculate the surface hardness of the specimen.

Two standard hardness tests are the Rockwell and Brinell tests. Other test methods included Sclera scope, surface abrasion testing, Vickers, and Tukon-Knoop.

TESTINGPROCESS FOR HARDNESS

In most tests, a steel ball with a diameter of 10 mm is utilized as an indenter. The standard loads are 250 kg, 500 kg, and 3000 kg.

Brinell hardness is determined by pressing a hard steel or carbide sphere of a specific diameter into a material's surface under a specific load and measuring the diameter of the indentation left behind. The Brinell hardness number, or simply the Brinell number, is derived by multiplying the weight in kilogram's by the actual surface area in square millimeters of the indentation. The result is a pressure measurement, however the units are rarely specified. mentioned.

Table3: HardnessValuesofComposites[InBHT]

Sample	Composition of composite Specimen	Trail1 (BHN)	Trail2 (BHN)	Trail3 (BHN)	Average
Sample 1	Al-950 gms + SiC-38 gms	65	65	64	64.66
Sample 2	Al-950 gms + SiC-76 gms	63	62	62	62.33
Sample 3	Al-950 gms + SiC-114 gms	78	80	82	80



Fig6:SamplesafterBrinellHardnessTest

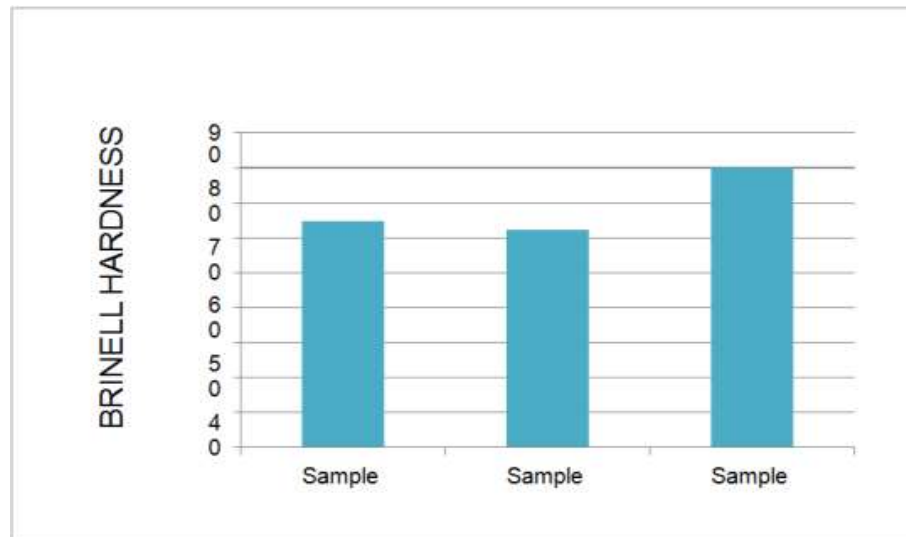


Fig 7: Graphical representation of Hardness Test values onCompositesamples.



Fig8:SamplesafterImpactTest

C) IMPACT TESTING

The ability of a substance to withstand a sudden impact is known as impact strength. The impact test was carried out in a Charpy impact test bed, as illustrated in the picture. The tests were carried out using an impact tester in accordance with "ASTM D256". The impact test specimen measured 10mm X 10mm X 55mm thick. As illustrated in the image, the specimen was positioned horizontally in the test bed. From the standard height, the pendulum was raised and made to strike the specimen.



Fig9: Impact test machine

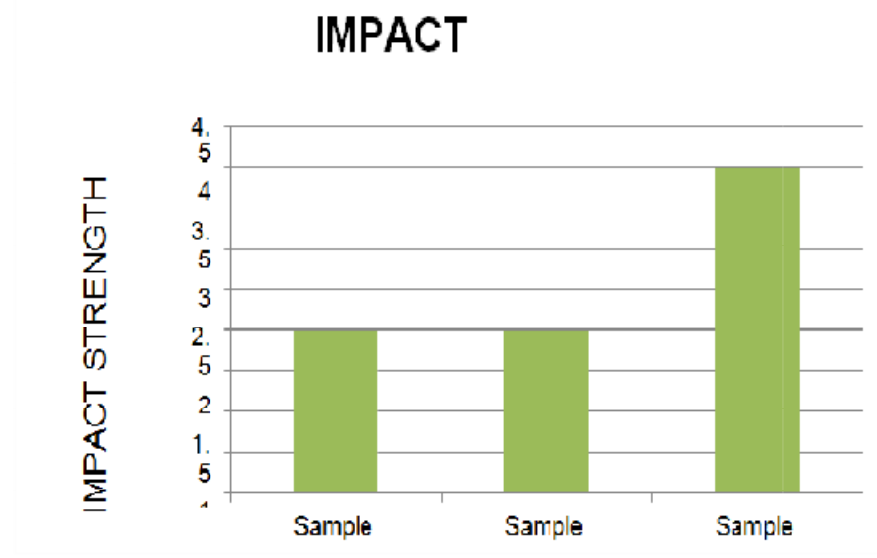


Fig 10: Graphical representation of impact Test values on Composite samples.

D) TEST FOR COMPRESSIONSTRENGTH

CompressiveThe ability of a material or structure to tolerate stressors that cause it to shrink in size, as opposed to tensile strength, which withstands loads that cause it to lengthen, is known as compressive strength. In material strength study, tensile strength, compressive strength, and shear strength can all be examined individually.

Some materials break at their compressive strength limit, while others deform permanently; thus, the compressive load limit can be defined as a particular degree of deformation. When constructing structures, compressive strength is a crucial consideration. Compressive strength measurements are influenced by the test method and measurement conditions.

Table4:Compressiontestvaluesoncomposite

SAMPLE	COMPOSITION OF COMPOSITE SPECIMEN	COMPRESSIVE STRENGTH
SAMPLE-1	Al-950 gms + SiC-38 gms	224 Mpa
SAMPLE-2	Al-950 gms + SiC-76 gms	266 Mpa
SAMPLE-3	Al-950 gms + SiC-114 gms	209 Mpa



Fig11:Samplesaftercompressiontest

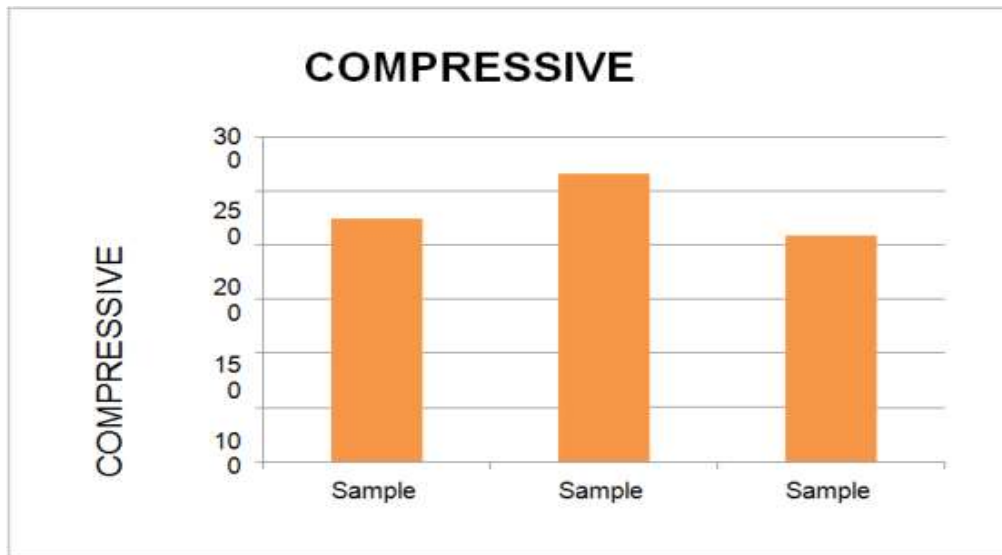


Fig 12: Graph for compression Test values on Composite samples.

WEAR RESISTANCE TEST

The term "wear resistance" is widely used to describe a material's anti-wear properties. The ASTM E122 standard was followed in the preparation of the test specimens. The scientific concept of wear resistance, on the other hand, is vague, and there is no standard method for determining it. Nonetheless, the inverse of the (relative) wear resistance is usually used to compute it.

Loss of bulk or volume Comparing the wear loss of a reference material to that of the researched material under the same testing conditions can also be used to compute relative wear resistance. In any case, the importance of a quantitative wear resistance number should be expressed properly.

E) WEAR TEST ON PIN-ON DISK MACHINE

A In a pin-on-disc wear tester, a pin is loaded against a flat rotating disc specimen, and the machine describes a circular wear path. The machine can be used to test the wear and friction properties of materials under pure sliding settings. The specimen can be a disc or a pin, with the counter face being the other. Different geometries of pins can be used. A ball composed of commercially available materials such as bearing steel, tungsten carbide, or alumina (Al_2O_3) can be used as a counter face, earning the term "ball-on-disc."

Table 5: Wear test values on composite

Sample no	Initial weight(g)	Final weight(g)	Abrasion loss(g)	Wear Rate (%)
1	5.1285	4.9521	0.1764	3.44
2	5.1028	4.9543	0.1485	2.91
3	5.1079	4.9393	0.1686	3.30

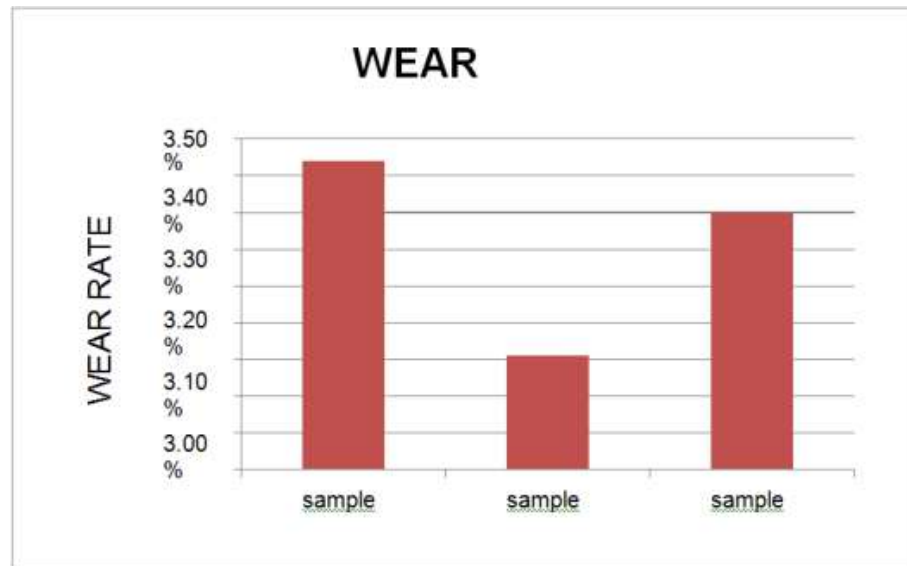


Fig13:Graph for wear Test values on Composite samples.

5. CONCLUSION

The current research looks at how to manufacture an aluminium alloy (8090) reinforced with SiC composite. The mechanical behaviour of the composite produced the following outcomes.

- ❖ Fabrication of a novel type of aluminium alloy with SiC reinforcement has proven successful.
- ❖ It was determined that sample 3's tensile strength (160 Mpa) is higher than samples 1 and 2. (48 and 93 Mpa). The abundant increase in tensile strength is due to applied tensile load transfer to the strongly bonded SiC reinforcements in Al matrix, increased dislocation density near the matrix-reinforcement interface, and grain refining strengthening effect, on the other hand. The tensile strengths of all three samples ranged from 48 to 170 MPa..
- ❖ The hardness of sample 3 of these composites is found to be higher, with a value of 80 BHN. The hardness of AMCs is increased by the presence of harder and better-bonded SiC particles in the Al matrix, which prevent dislocation migration. The hardness of three samples ranges from 60 to 85 BHN.
- ❖ Sample 2 of these composites has a compressive strength of 266 MPa, which is determined to be greater. This is owing to the inclusion of SiC, a strong material, to Aluminum, which needs a high level of strength in order to compress the AMCs..
- ❖ However, due to the aluminium alloy's adhesion to the SiC, sample 3 has a higher impact strength. The impact strength of all three samples is around 2-4 joules.
- ❖ However, sample 2's wear resistance analysis is satisfactory. This is because the softer Al matrix wears away first from the sample's surface during the wear test, leaving the hard SiC particles on the worn surface. The exposed SiC particles shield the Al matrix from further wear.

SCOPE OF FUTURE WORK

Future researchers will have plenty of opportunities to look at this topic. This research could be broadened to look into other tribological and thermal characteristics of this composite, such as abrasion, microstructure, and thermal conductivity. Various aspects of such composites can be investigated, such as the use of various potential fillers in the building of metal matrix composites and the evaluation of their mechanical and erosion behaviour, as well as the experimental results.

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